

Infrasonic Evolution

MTV Workshop, 2020

10 March 2020

Milton Garces University of Hawaii





Year 1: Introduction and Motivation

TA3.1.Y1. Infrasound

- Address diversity of emplacement conditions and delivery systems for low, high, and nuclear explosives.
- Apply new blast pulse scaling and parametrization developed during CVT to airborne and near-surface stationary, subsonic, supersonic, and hypersonic explosive sources.

Students on Infrasound Thrust (see Falcon 9 study in their poster)

- Kei Takasawa
- Jonathan Tobin

Lab Collaborations

- LLNL: Keehoon Kim, Jessie Gaylord and Steven Magana-Zook.
- SNL: Daniel Bowman. Airborne platforms.
- NNSS: Cleat Zeiler. LANL: Jim Smith. Source physics and processes.
- INL: David Chichester. Pursue new signatures and methods.







Blasts: from kilotons to tons

Modernize geophysical sensing systems and methods that can be deployed globally to effectively monitor explosive events and fuel cycle activity.







High Explosive NE Proxies at NNSS



Supersonic Propellant Blast Proxy





http://redvox.io/@/a3a8

Launch and photo by D. Bowman.

Emerging Collection Platforms







Explosion Source (K. Kim, LLNL).

Canonical Waveform (Analytic) from Garces (2019)



- Traditional Mission
- HE Proxy for NE
- Static source and receiver positions
- Transition from CVT to MTV

Kim, K., A. J. Rodgers, M. A. Garces, S. Myers (2019). An Empirical Acoustic Source Model for Chemical Explosions in Air. Amer. Geophys. Union Fall Meeting, SF, CA, 9-13 Dec. Also presented in Jordan ITW (2019)





Stratospheric Blast: Falcon 9 Burn at Mach 2

Moving Source. Deflagration not the same as a detonation. Supersonic; shocks expected. Source at ~20 km (stratosphere). Crowdsourced open data: <u>http://redvox.io/@/f8cf</u>











Blasts: Falcon 9 at Mach 2

Open data: <u>http://redvox.io/@/f8cf</u>







Twin Blast Signatures: Falcon 9 at Mach 2

Open data: <u>http://redvox.io/@/f8cf</u>

Short-term Fourier Transform







Constant Q Transform

Falcon 9 RP-1 Deflagration at Mach 2

Moving source; NOT a point source. Hypothesis: Shock tube.



Table 2-1: Falcon dimensions and characteristics

Characteristic	First Stage Core	Second Stage
Structure		
Height	70 m (229 ft) (including both stages, interstage and fairing)	
Diameter	3.66 m (12 ft)	3.66 m (12 ft)
Туре	LOX tank – monococque;	LOX tank – monococque
	Fuel tank – skin and stringer	Fuel tanks – skin and stringer
Material	Aluminum lithium skin; aluminum domes	
Propulsion		
Engine type	Liquid, gas generator	Liquid, gas generator
Engine designation	Merlin 1D (M1D)	MVac
Engine designer	SpaceX	SpaceX
Engine manufacturer	SpaceX	SpaceX
Number of engines	9	1
Propellant	Liquid oxygen/kerosene (RP-1)	Liquid oxygen/kerosene (RP-1)
Thrust (stage total)	7,686 kN (sea level) (1,710,000 lbf)	981 kN (Vacuum) (220,500 lbf)







Stratospheric Collection (with D. Bowman, SNL)



- 2281691807 400 17 300 SNR (dB) ₽ 200 100 400 33 300 P² (dB) ₽ 200 100 42 $\times 10^4$ 1 Counts 0 17:29 17:35 17:30 17:31 17:32 17:34 17:36 17:37 17:33 UTC 2018/10/10
- Light aircraft characterized by Doppler-shifted signatures with fundamental and overtones
- Moving source, moving receiver

Trajectory data on 2018/10/10 at 17:28:06 UTC Lat, Lon, Altitude (m): 34.99350, -104.76358, +19,536 m Temperature, Pressure: -31.1C, 5965 Pa





Conclusions and Ongoing Work

- Garces (2019) canonical blast pulse validated by LLNL explosion data. Transitioning from CVT to MTV.
- Supersonic blasts: 20200119 SpaceX Dragon case study.
- Theoretical full-wave equations for moving subsonic acoustic sources and receivers with a reflecting boundary rederived, coded, and validated against observations. Collaboration with SNL for collection with airborne platforms.
- Nevada National Security Site (NNSS) blasts scheduled for 2020.
- New deployments in planning stage with INL.



